Shearing Stresses in Beams and Thin-Walled Members

Part II

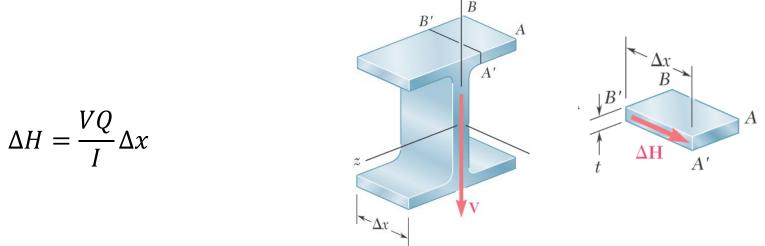
## **Shearing Stresses In Thin-walled Members**



• The longitudinal shear and horizontal shear per unit length (shear flow) are used in this section to calculate both the shear flow and the average shearing stress in thin-walled members such as the flanges of wide-flange beams, box beams, or the walls of structural tubes

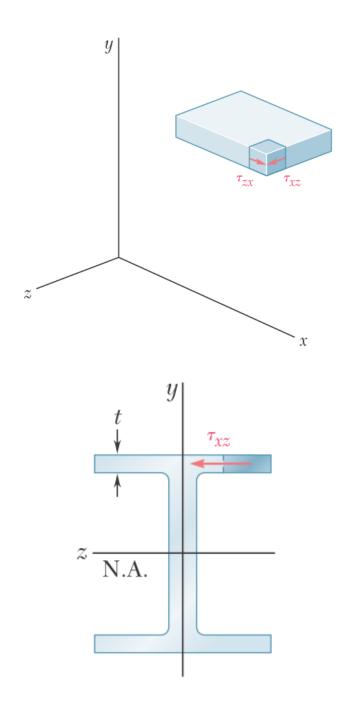
$$\Delta H = \frac{VQ}{I} \Delta x \qquad \qquad q = \frac{\Delta H}{\Delta x} = \frac{VQ}{I}$$

• Consider a segment of length  $\Delta x$  of a wide-flange beam where V is the vertical shear in the transverse section shown. Detach an element ABB'A' of the upper flange. The longitudinal shear  $\Delta H$  exerted on that element can be obtained as



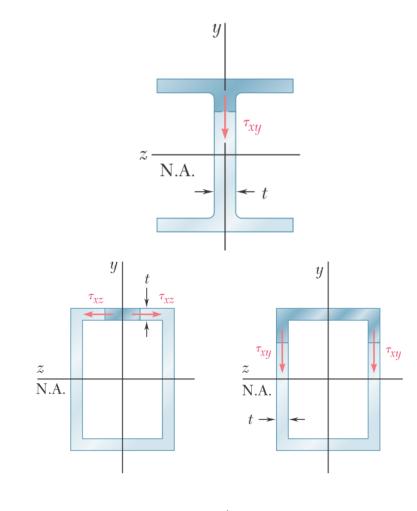
• Dividing  $\Delta H$  by the area  $\Delta A = t\Delta x$  of the cut, the average shearing stress exerted on the element is the same expression obtained in the previous course for a horizontal cut:

$$\tau_{ave} = \frac{VQ}{It}$$



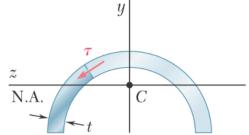
• Note that  $\tau_{ave}$  now represents the average value of the shearing stress  $\tau_{zx}$  over a vertical cut, but since the thickness t of the flange is small, there is very little variation of  $\tau_{zx}$  across the cut.

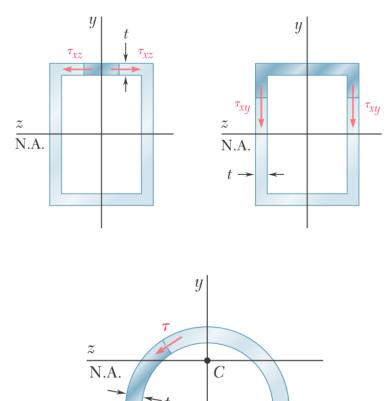
• Recalling that  $\tau_{zx} = \tau_{xz}$ , the horizontal component  $\tau_{xz}$  of the shearing stress at any point of a transverse section of the flange can be obtained from  $\tau_{ave} = \frac{VQ}{It}$ , where Q is the first moment of the shaded area about the neutral axis.



• A similar result was obtained for the vertical component  $\tau_{xy}$  of the shearing stress in the web.

•  $\tau_{ave} = \frac{VQ}{It}$  can be used to determine shearing stresses in box beams, half pipes, and other thin-walled members, as long as the loads are applied in a plane of symmetry.



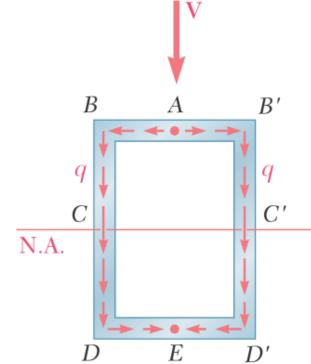


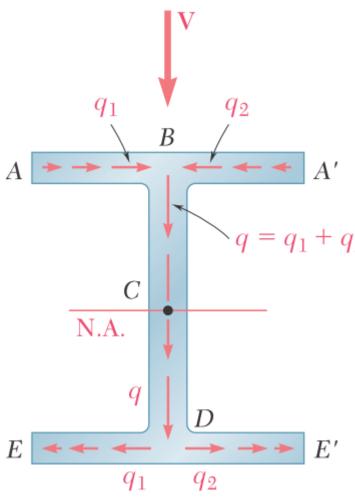
• In each case, the cut must be perpendicular to the surface of the member, and  $\tau_{ave} = \frac{VQ}{It}$  will yield the component of the shearing stress in the direction tangent to that surface.

• The other component is assumed to be equal to zero, because of the proximity of the two free surfaces.

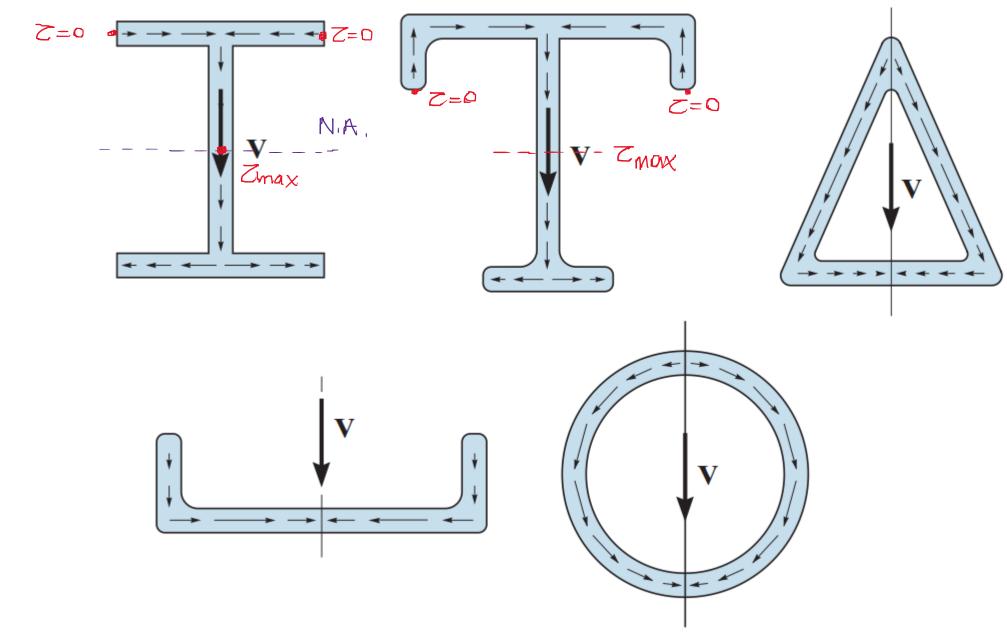
- Comparing  $q = \frac{VQ}{I}$  and  $\tau_{ave} = \frac{VQ}{It}$ , the product of the shearing stress  $\tau$  at a given point of the section and the thickness t at that point is equal to q.
- Since *V* and *I* are constant, *q* depends only upon the first moment *Q* and easily can be sketched on the section.
  - For a box beam, q grows smoothly from zero at A to a maximum value at C and C' on the neutral axis and decreases back to zero as E is reached.

There is no sudden variation in the magnitude of q as it passes a corner at B, D, B', or D', and the sense of q in the horizontal portions of the section is easily obtained from its sense in the vertical portions (the sense of the shear V).





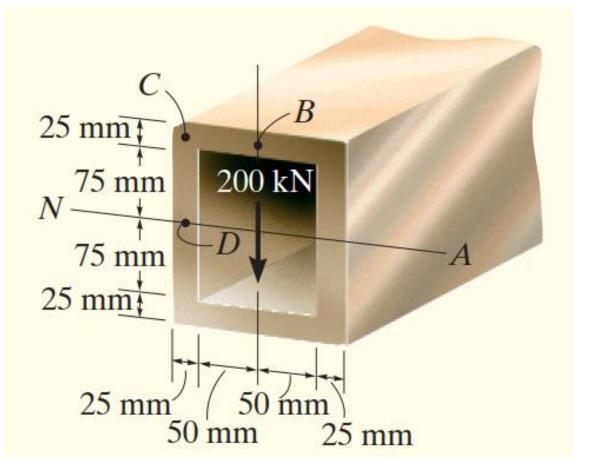
- In a wide-flange section, the values of q in portions AB and A'B of the upper flange are distributed symmetrically.
- At B in the web, q corresponds to the two halves of the flange, which must be combined to obtain the value of q at the top of the web.
- After reaching a maximum value at C on the neutral axis, q decreases and splits into two equal parts at D, which corresponds at D to the two halves of the lower flange.
  - The shear per unit length q is commonly called the shear flow and reflects the similarity between the properties of q just described and some of the characteristics of a fluid flow through an open channel or pipe.



#### **Shear Flow in Common Thin-Walled Members**

#### Example

The thin-walled box beam in figure is subjected to a shear of 200 kN. Determine the variation of the shear flow throughout the cross section.



### **Solution**

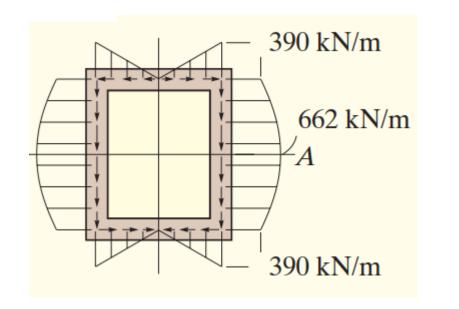
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#### **Solution**

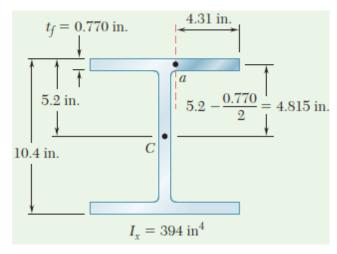
# $Q_{D} = \Sigma \overline{y}' A' = 2 \left[ \left( \frac{0.0875 \text{ m}}{2} \right) (0.025 \text{ m}) (0.0875 \text{ m}) \right]$ + [0.0875 m](0.125 m)(0.025 m) = 0.4648(10^{-3}) m^{3} $q_{D} = \frac{1}{2} \left( \frac{VQ_{D}}{I} \right) = 662 \text{ kN/m}$

At point D.



Using these results, and the symmetry of the cross section, the shear-flow distribution is

#### Sample Problem 6.3



Knowing that the vertical shear is 50 kips in a W10 X 68 rolled-steel beam, determine the horizontal shearing stress in the top flange at a point a located 4.31 in. from the edge of the beam. The dimensions and other geometric data of the rolled-steel section are given in Appendix C.

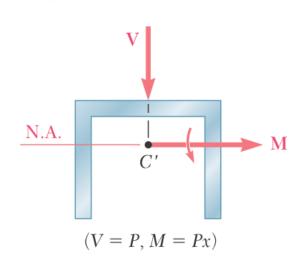
 $Q = (4.31 \text{ in.})(0.770 \text{ in.})(4.815 \text{ in.}) = 15.98 \text{ in}^3$ 

$$\tau = \frac{VQ}{It} = \frac{(50 \text{ kips})(15.98 \text{ in}^3)}{(394 \text{ in}^3)(0.770 \text{ in.})} \to \tau = 2.63 \text{ ksi}$$

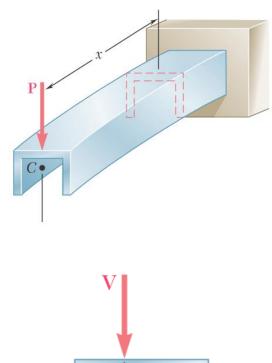
## Unsymmetric Loading Of Thin-walled Members and Shear Center

- Our analysis of the effects of transverse loadings has been limited to members possessing a vertical plane of symmetry and to loads applied in that plane.
- The members were observed to bend in the plane of loading, and in any given cross section, the bending couple M and the shear V were found to result in normal and shearing stresses:

$$\sigma_x = -\frac{My}{I}$$
 and  $\tau_{ave} = \frac{VQ}{It}$ 



P

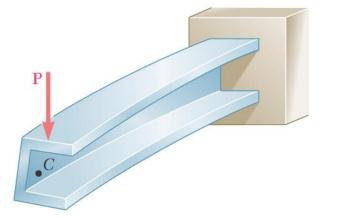


N.A. V = P, M = Px

- In this section, the effects of transverse loads on thin-walled members that do not possess a vertical plane of symmetry are examined.
- Assume that the channel member has been rotated through 90° and that the line of action of P still passes through the centroid of the end section.
- The couple vector M representing the bending moment in a given cross section is still directed along a principal axis of the section, and the neutral axis will coincide with that axis.

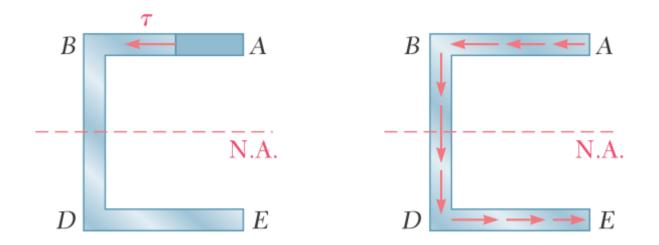
•  $\sigma_{\chi} = -\frac{My}{I}$  can be used to compute the normal stresses in the section. However,  $\tau_{ave} = \frac{VQ}{It}$  cannot be used to determine the shearing stresses, since this equation was derived for a member possessing a vertical plane of symmetry.

• Actually, the member will be observed to bend and twist under the applied load, and the resulting distribution of shearing stresses will be quite different from that given by  $\tau_{ave} = \frac{VQ}{It}$ .

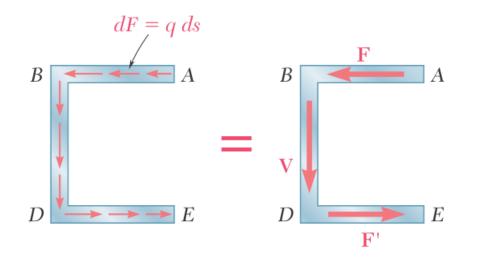


 Is it possible to apply the vertical load P so that the channel member of figure will bend without twisting?

• If so, where should the load P be applied?



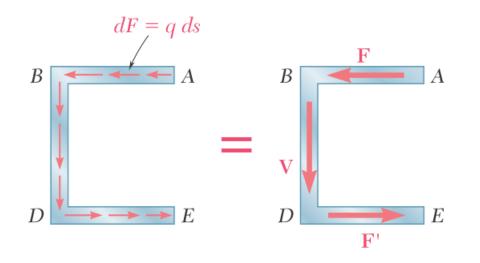
• If the member bends without twisting, the shearing stress at any point of a given cross section can be obtained from  $\tau_{ave} = \frac{VQ}{It}$ , where Q is the first moment of the shaded area with respect to the neutral axis and the distribution of stresses is as shown with  $\tau = 0$  at both A and E.



• The shearing force exerted on a small element of cross-sectional area dA = tds is  $dF = \tau dA = \tau tds$  or dF = qds, where q is the shear flow  $q = \tau t = \frac{VQ}{I}$ . The resultant of the shearing forces exerted on the elements of the upper flange AB of the channel is a horizontal force F of magnitude

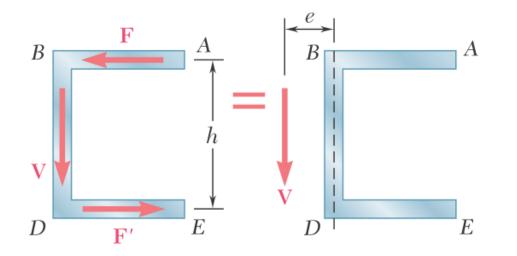
$$F = \int_{A}^{B} q ds$$

 Because of the symmetry of the channel section about its neutral axis, the resultant of the shearing forces exerted on the lower flange DE is a force F' of the same magnitude as F but of opposite sense.



• The resultant of the shearing forces exerted on the web BD must be equal to the vertical shear V in the section:

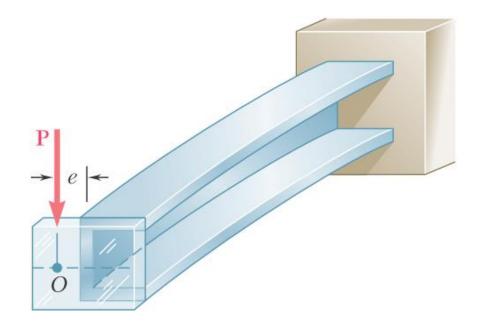
$$V = \int_{B}^{D} q ds$$



• The forces F and F' form a couple of moment  $F \times h$ , where h is the distance between the center lines of the flanges AB and DE. This couple can be eliminated if the vertical shear V is moved to the left through a distance e so the moment of V about B is equal to  $F \times h$ . Thus,  $V \times e = F \times h$  or

$$e = \frac{Fh}{V}$$

- When the force P is applied at a distance e to the left of the center line of the web BD, the member bends in a vertical plane without twisting.
- The point O where the line of action of P intersects the axis of symmetry of the end section is the shear center of that section.



- In the case of an oblique load P, the member will also be free of twist if the load P is applied at the shear center of the section.
- The load P then can be resolved into two components P<sub>z</sub> and P<sub>y</sub> neither of which causes the member to twist.

